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# System Optimization of Gasdynamic Lasers Computer Program User's Manual

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L. J. Otten III, R. C. Saunders III, and S. J. Morris

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# SYSTEM OPTIMIZATION OF GASDYNAMIC LASERS

## COMPUTER PROGRAM USER'S MANUAL

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### SUMMARY

This report contains the user's manual for the computer program described in reference 1. Gasdynamic laser systems or subsystems can be optimized by using this manual in conjunction with reference 1. This manual assumes familiarity with gasdynamic laser systems and with large-scale computer models.

### PROGRAM DESCRIPTION

Necessarily, the computer code that performs this analysis is large and complex. In its present form, it has 32 subroutines and in excess of 6600 FORTRAN statements, of which perhaps 10-20% are comments. The code is written in a dialect of FORTRAN and was originally intended for the Control Data Corporation 6600/7600 computers. Hence, it uses a few language features and programming techniques that are unique to that machine. It has recently been converted to other machines, however, including the International Business Machines 360/370 computers, so its machine-dependence is minimal. A single analysis pass requires from 3 to 5 sec on the CDC 7600, and a typical optimization may require 300 sec or more. The program requires 106,000<sub>8</sub> words of memory on the CDC 7600 and operates without overlay or segments on this system. No external storage such as magnetic tape is required.

The program consists of five modules, as shown in figure 1. The control module performs certain input and output functions, maintains the flow of information between modules, and calls the other modules at the appropriate time. The optimization module (CONMIN) is used to minimize or maximize some features of the system. The remaining three modules (GAIN, FLAME, and WTVOL) are the physical models and calculate the laser performance, combustion information, and weight/volume results, respectively. The modules and the FORTRAN subroutines associated with each of them are summarized in table 1.

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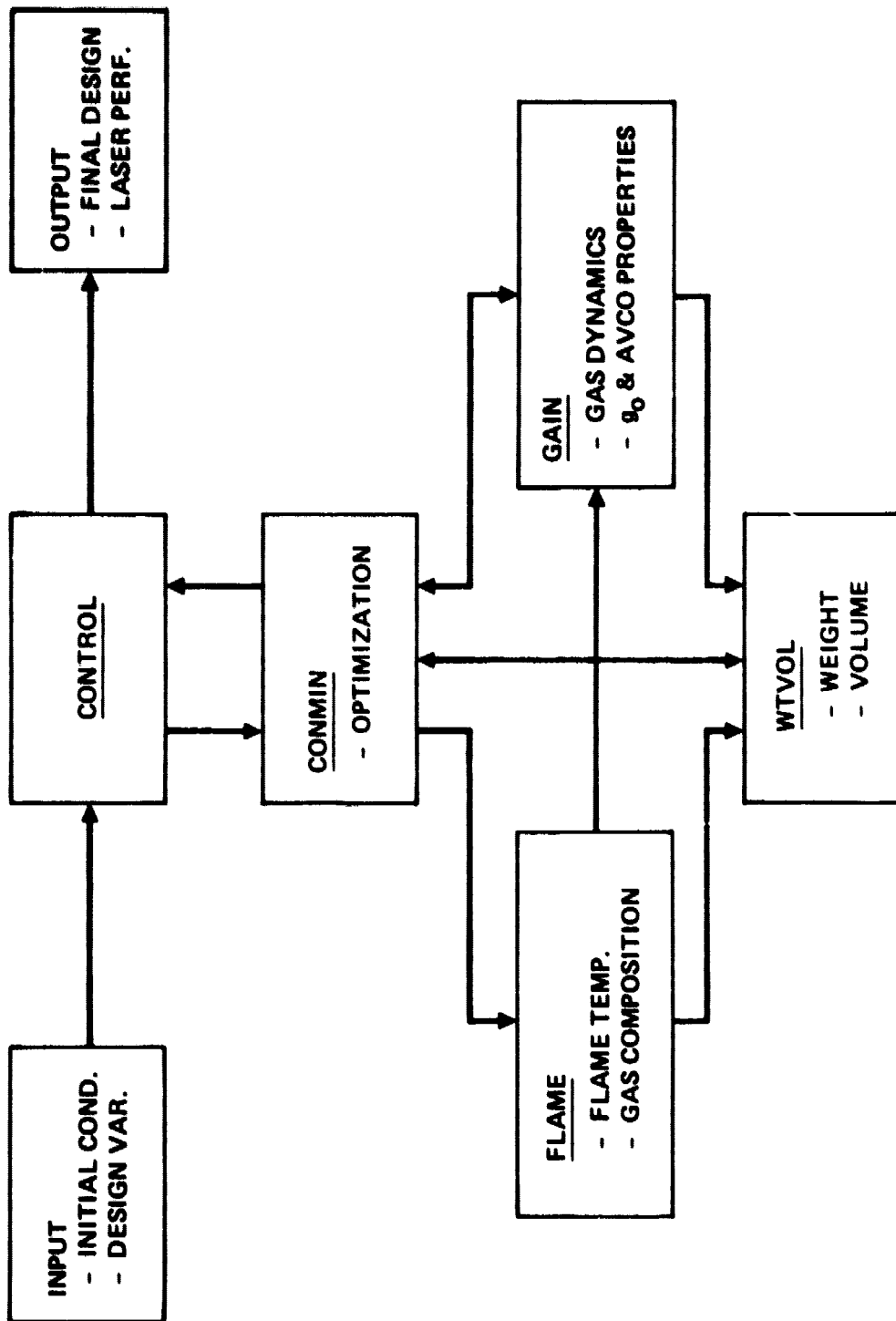


Figure 1.- Program structure.

TABLE 1.- PROGRAM MODULES AND SUBROUTINES

Module	Function	Associated subroutines
CONTROL	I/O control; communication	COPE2* COPE01 COPE02 COPE03 COPE04 COPE05 COPE06 COPE07
CONMIN	Optimization	CONMIN* CNMN01 CNMN02 CNMN03 CNMN04 CNMN05 CNMN06 CNMN07 CNMN08
GAIN	Laser gasdynamics and kinetics; gain, available energy	CYCLE* CHAR DERFC FLUX TAUSUB RUNGEX EV NOZZLE CONTOUR MACH GAMMA TABNOZ AVG
FLAME	Combustion products; flame temperature	FLAME*
WTVOL	Weight; volume	WTVOL*

\*Denotes entry point.

Because this program was developed over an extended period of time, each of the physical model modules (GAIN, FLAME, and WTVOL) was initially written to run on its own. Therefore, each of these modules was required to read the information it needed, execute, and write the output on its own. For simplicity, this structure was retained in the integrated program. In addition to making module development easier, the use of this structure permits easy substitution of any of the modules. Execution sequence is controlled by the CONTROL module.

Communication between the modules and especially with the control module is handled by a single labeled COMMON block, GLOBCM. A parameter's position in this block is used to identify its role in the sensitivity analysis or optimization process. The contents of this block, called the global COMMON block, are shown in table 2. In the version described here, only the first 35 elements of this block are used; the remaining 1465 are available for future use or can be deleted if memory space is at a premium.

TABLE 2.- GLOBAL CATALOG

Location	FORTTRAN name	Definition	Units	Defined by <sup>a</sup>	Used by <sup>a</sup>
1	TSTAG	Combustor stagnation temperature	K	1, 2	1, 3
2	PSTAG	Combustor stagnation pressure	atm	1	1, 3
3	THROAT	Cavity nozzle throat height	mm	1	1, 3
4	AEXIT	Cavity nozzle area ratio	---	1	1
5	XC	Mole fraction of CO <sub>2</sub>	---	1, 2	1
6	XH2	Mole fraction of H <sub>2</sub> O	---	1, 2	1
7	XCO	Mole fraction of CO	---	1	1
8	XHE	Mole fraction of H <sub>2</sub>	---	1	1
9	ELSUB	Length of subsonic portion of nozzle	cm	1	1
10	ELNOZ	Length of supersonic portion of nozzle	cm	1	1
11	C	Length of 1st section of nozzle	cm	1	1
12	RAD	Length of 2nd section of nozzle	cm	1	1
13	THETA1	1st nozzle angle	deg	1	1
14	THETA2	2nd nozzle angle	deg	1	1
15	BIG	Maximum peak gain	%/cm	1	1
16	PAVG	Average available specific energy	kJ/lb	1	1, 3
17	CHI	Ratio of moles of carbon to moles of C-N-O-H fuel	---	2	2

<sup>a</sup>1 - GAIN  
2 - FLAME  
3 - WTVOL

TABLE 2.- Concluded

Location	FORTTRAN name	Definition	Units	Defined by <sup>a</sup>	Used by <sup>a</sup>
18	Q	Moles of excess oxidizer	---	2	2
19	R	Ratio of moles of N <sub>2</sub> to moles of C-H-O-H fuel (OX/AIR)	---	2	2
20	S	Ratio of moles of excess CO <sub>2</sub> to moles of C-N-O-H fuel	---	2	2
21	DELNX	Additional oxidizer temp. due to preheat	K	2	2
22	DELN2	Additional makeup N <sub>2</sub> temp. due to preheat	K	2	2
23	QFRAC	Mole fraction of N <sub>2</sub> that is preheated	---	2	2
24	WTO	System weight (old bench)	lb	3	3
25	VT	System volume (old bench)	ft <sup>3</sup>	3	3
26	WTO2	System weight (new bench)	lb	3	3
27	VT2	System volume (new bench)	ft <sup>3</sup>	3	3
28	B	Cavity/mirror height	in.	3	1, 3
29	ARCAV	Cavity aspect ratio	---	3	3
30	P	Static pressure at end of cavity	atm	1	3
31	EM	Mach number at end of cavity	---	1	3
32	PGAM	Cavity process $\gamma$	---	1	3
33	ETA	Diffuser efficiency (times normal shock)	---	3	3
34	<del>P0</del> -P0MIN	Actual minus required pres- sure at end of diffuser ( $>0$ for recovery)	psi	3	3
35	ETAN2	Ratio of moles of N <sub>2</sub> to moles of N <sub>2</sub> in air (OX/AIR)	---	2	2

## PROGRAM USAGE

The program input is divided into five sections. The first section is CONTROL and CONMIN input. The second section selects which options among the physical models will be used. The remaining sections are the inputs for the physical model modules. The first two sections of input are mandatory for any problem. The remaining sections of input depend on the particular problem under investigation.

This section provides a card-by-card description of the input and presents a sample. The format for each card is given in parenthesis after the contents, and a description of the contents is given below it.

#### A. CONTROL and CONMIN Input (Mandatory)

Card 1: TITLE (20A4)

TITLE - any alphanumeric title, up to 80 characters.

Card 2: NCALC, NDV, NSV, N2VAR, IPNPUT, IPSENS, IP2VAR (715)

NCALC - calculation control.

- 0: read input and stop. All other control input is optional.
- 1: cycle through program once. All other control input is optional.
- 2: do optimization. Additional control input of Cards 3-10 is required. All other control input is optional.
- 3: do sensitivity analysis. Additional control input of Cards 11-14 is required. All other control input is optional.
- 4: do optimum sensitivity. Additional control input of Cards 3-14 is required. All other control input is optional.
- 5: do two-variable analysis (all combinations of two variables). Additional control input of Cards 15-18 is required. All other control input is optional.
- 6: do optimum two-variable analysis. Additional control input of Cards 3-10 and 15-18 is required. All other control input is optional.

NDV - number of design variables in optimization or optimum sensitivity.

NSV - number of sensitivity variables.

N2VAR - number of objective functions in a two-variable analysis.

IPNPUT - input print control.

- 0: print card images plus formatted print.
- 1: formatted print only.
- 2: no print of input.

IPSENS - sensitivity print control.

0: no detailed print each sensitivity analysis.

1: detailed print each sensitivity analysis.

(DEFAULT: 0)

IP2VAR - two-variable analysis print control.

0: no detailed print each two-variable analysis.

1: detailed print each two-variable analysis.

(DEFAULT: 0)

- - - - - OPTIMIZATION - - - - -

Card 3: IPRINT, ITMAX, ICNDIR, NSCAL, ITRM, LINOBJ, NACMX1, NFDG (815)

IPRINT - optimization print control.

0: no print.

1: print initial and final optimization information.

2: 1 plus function value and design variable values at each iteration.

3: 2 plus constraint values, direction vector, and move parameter at each iteration.

4: 3 plus gradient information.

ITMAX - maximum number of optimization iterations (DEFAULT = 10)

ICNDIR - conjugate direction restart parameter (DEFAULT = 3)

NSCAL - scaling frequency. Design variables are scaled every NSCAL iteration. (DEFAULT = 0, no scaling)

ITRM - number of successive iterations which must satisfy convergence criteria (see Card 5) before optimization is terminated.

(DEFAULT = 3)

LINOBJ - linear objective function identifier.

0: objective function is nonlinear function of design variables.

1: objective function is linear function of design variables.

(DEFAULT = 0)

NACMX1 - number of anticipated active constraints plus 1.

(DEFAULT: NDV + 2)

NFDG - finite difference gradient identifier.

0: gradients computed by finite difference.

1: gradient of objective computed analytically, gradients of constraints computed by finite difference.

2: all gradients computed analytically.

(DEFAULT = 0)

Card 4: FDCH, FDCHM, CT, CTMIN, CTL, CTLMIN, THETA, PHI (8F10.2)

FDCH - relative finite difference size for gradient calculation.

(DEFAULT - 0.01)

FDCHM - absolute finite difference size for gradient calculation.

(DEFAULT - 0.001)

CT - constraint thickness parameter.

(DEFAULT - 0.05)

CTMIN - minimum absolute value of CT

(DEFAULT - 0.004)

CTL - constraint thickness parameter for linear and side constraints.

(DEFAULT - 0.01)

CTLMIN - minimum absolute value of CTL.

THETA - mean value of push-off factor.

(DEFAULT - 1.0)

PHI - participation coefficient

(DEFAULT = 5.0)

Card 5: DELFUN, DABFUN (2F10.2)

DELFUN - minimum relative change in objective function. Used to terminate optimization.

(DEFAULT = 0.001)

DABFUN - minimum absolute change in objective function. Used to terminate optimization.

(DEFAULT = 0.001)

Card 6: NDVTOT, IOBJ, SCNOPT

(215,F10.2)

NDVTOT - total number of variables linked to the design variables. NDVTOT must be greater than or equal to NDV (see Card 2). This option allows two or more parameters to be assigned to a single design variable. The value of each parameter is the value of the design variable times a multiplier, which may be different for each parameter.

(DEFAULT: NDV)

IOBJ - global catalog location of objective function for optimization.

SCNOPT - optimization control.

-1: minimize objective function.

+1: maximize objective function

Card 7: VLB(I), VUB(I), X(I)

(3F10.2)

VLB(I) - lower bound on Ith design variable.

VUB(I) - upper bound on Ith design variable.

X(I) - initial value of Ith design variable (if  $\neq 0$ , it overrides the value read later)

NOTE: NDV (see Card 2) of these cards are required.

Card 8: NDSGN(I), IDSGN(I), AMULT(I)

(215,F10.2)

NDSGN(I) - number associated with the Ith design variable.

IDSGN(I) - global catalog location of Ith design variable.

AMULT(I) - constant multiplier on the Ith design variable.

NOTE: IDV (see Card 2) of these cards are required.

Card 9: NCONS

(15)

NCONS - number of constraint sets in an optimization.



Card 10: ICON(I), JCON(I), LCON(I), BLU(I,1), BLU(I,2), BLU(I,3) (3I5,3F10.2)

ICON(I) - first global catalog location of the Ith constraint variable.

JCON(I) - last global catalog location of the Ith constraint variable.

(DEFAULT = ICON(I))

LCON(I) - constraint identifier for the Ith constraint variable.

1: linear constraint

0: nonlinear constraint

BLU(I,1) - lower bound on Ith constraint variable.

BLU(I,2) - upper bound on Ith constraint variable.

BLU(I,3) - scaling factor on Ith constraint variable.

(DEFAULT = (BLU(I,1) + BLU(I,2))/2.)

NOTE: NCONS (see Card 9) of these cards are required.

- - - - - SENSITIVITY ANALYSIS - - - - -

Card 11: NSOBJ (15)

NSOBJ - number of objective functions for sensitivity analysis

Card 12: NSENSZ(I), I = 1, NSOBJ (1615)

NSENSZ(I) - global catalog location of Ith sensitivity objective function.

Card 13: ISENS(I), NSENS(I) (215)

ISENS(I) - global catalog location of Ith sensitivity variable.

NSENS(I) - number of values of Ith sensitivity variable.

Card 14: SENS (I, J)

SENS (I, J) - values of Ith sensitivity variable (J = 1 is nominal value)

NOTE: NSV (see Card 2) of Cards 13-14 are required.

- - - - - TWO-VARIABLE FUNCTION SPACE - - - - -

Card 15: N2VX, M2VX, N2VY, M2VY (4I5)

N2VX - global catalog location of X-variable.

M2VX - number of values of X-variable.

N2VY - global catalog location of Y-variable.

M2VY - number of values of Y-variable.

Card 16: N2VZ(I), I = 1, N2VAR (5I5)

N2VZ - global catalog location of Ith objective. N2VAR is defined by Card 2.

Card 17: XM2V(I), I = 1, M2VX (8F10.2)

XM2V(I) - Ith value of X-variable. M2VX is defined by Card 15.

Card 18: YM2V(I), I = 1, M2VY (8F10.2)

YM2V(I) - Ith value of Y-variable. M2VY is defined by Card 15.

Card 19: END

The word 'END' must appear here at the end of the control input.

#### B. OPTION Input (Mandatory)

A single card in NAMELIST format is used for inputting to this section.

Card 20: WEIGHT, BURNER, LASER, VLCAV, DEBUG (5L, NAMELIST "OPTION")

WEIGHT - is the WTVOL module to be used?

(DEFAULT = F)

BURNER - is the FLAME module to be used?

(DEFAULT = F)

LASER - is the GAIN module to be used?

(DEFAULT = T)

WT.CAV - is the cavity calculation to be carried 2 mirror diameters downstream? If not, the input value on Card 29 will be used. It is valid only if WEIGHT = T.

(DEFAULT = F)

DEBUG - write debug output?

(DEFAULT = F)

NOTE: The WTVOL module cannot be run alone (i.e., BURNER = F, LASER = F, WEIGHT = T), since certain parameters normally supplied by GAIN would not be available. FLAME and GAIN may be run alone.

### C. WTVOL Input

A single card in NAMELIST format is used for inputting this data. This card should be present only if WEIGHT = T (see Card 20).

Card 21: TP, TF, B, EMEXIT, PØEXIT, ETA, EFF, N, NOMISC, NOVALV (7F,1I,2L, NAMELIST "SIZE")

TP - power out of laser cavity, in watts.

(DEFAULT =  $5 \times 10^5$ )

TF - shot length, in seconds.

(DEFAULT = 2.0)

B - nozzle/mirror height, in inches.

(DEFAULT = 5.0)

EMEXIT - Mach number at the diffuser exit.

(DEFAULT = 0.9)

PØEXIT - total pressure at the diffuser exit, in PSI.

(DEFAULT = 14.7)

ETA - diffuser efficiency, in multiples of normal shock efficiency.

(DEFAULT = 0.5)

EFF - cavity extraction efficiency.

(DEFAULT = 1.0)

N - number of shots.

(DEFAULT = 15)

NOMISC - is "miscellaneous" weight to be omitted from the total system weight?

(DEFAULT = F)

NOVALV - is "valve" weight to be omitted from the total system weight?

(DEFAULT = F)

#### D. FLAME Input

A single card in NAMELIST format is used for inputting this data. This card should be present only if BURNER = T (see Card 20).

Card 22: CHI, Q, R, S, ETAN2, QFRAC, KFUEL, KOXID  
(SF, 2I, NAMELIST "PROPER")

CHI - moles of C to moles of C-N-O-H.

(DEFAULT = 0.0)

Q - moles of excess air (if air is oxidizer).

(DEFAULT = 0.0)

R - moles of  $N_2$  to moles of C-N-O-H (if oxidizer is not air).

S - moles of excess  $CO_2$  to moles of C-N-O-H.

(DEFAULT = 0.0)

ETAN2 - moles of  $N_2$  to moles of  $N_2$  in air (if oxidizer is air).

(DEFAULT = 1.0)

DELNZ - additional oxidizer temperature due to preheating, in degrees Kelvin.

(DEFAULT = 0.0)

DELN2 - additional makeup nitrogen temperature due to preheating, in degrees Kelvin.

(DEFAULT = 0.0)

QFRAC - fraction of makeup nitrogen at temperature DELN2.

(DEFAULT = 1.0)

KFUEL - fuel choice.

- 1: carbon monoxide (CO)
- 2: cyanogen ( $C_2N_2$ )
- 3: carbon and benzene ( $C + C_6H_6$ )
- 4: carbon (C)
- 5: benzonitrile ( $C_7H_5N$ )

KOXID - oxidizer choice.

- 1: oxygen ( $O_2$ )
- 2: nitrous oxide ( $N_2O$ )
- 3: nitrogen tetroxide ( $N_2O_4$ )
- 4: air

#### E. GAIN Input

This input section retains the format from the original code and, hence, a number of superfluous or redundant cards is still necessary, chiefly for (now nonexistent) plotting. Likewise, some of the input described below is for features of the code that have little utility in many problems, notably the restart capability. This entire section should be omitted if LASER = F (see Card 20).

Card 23: LASTPLT (11)

LASTPLT - an inactive plot parameter. It should be set to 1.

Card 24: IDO, TITLE (11, 7A10)

IDO - an inactive plot parameter. It should be set to 2.

TITLE - any alphanumeric title, up to 70 characters.

Card 25: TITLE2

TITLE2 - any alphanumeric title, up to 80 characters.

Card 26: IPLOT (11)

IPLOT - an inactive plot parameter. It should be set to 3.

Card 27: ITEST (11)

ITEST - an inactive flag. It should be set to 0.

Card 28: COMNT1 (8A10)

COMNT1 - any alphanumeric title up to 80 characters.

Card 29: TSTAG, PSTAG, AEXIT, THROAT, XMAX, ANGLE, FACTOR (7F10.0)

TSTAG - combustor stagnation temperature, in degrees Kelvin.  
If BURNER = T (see Card 20) this value is ignored.

PSTAG - combustor stagnation pressure, in atmospheres.

AEXIT - nozzle area ratio.

THROAT - nozzle throat height, in millimeters.

XMAX - maximum downstream distance over which cavity conditions are computed, in centimeters. If VLCAV = T and WEIGHT = T (see Card 20), this value is ignored.

ANGLE - cavity divergence angle, in degrees.

FACTOR - nozzle friction factor ( $\approx 1/2 \rho u^2$ )

Card 30: XC, XH2, XCO, XHE, JJ (4F10.0,12)

XC - mole fraction of  $\text{CO}_2$ . If BURNER = T (see Card 20), this value is ignored.

XH2 - mole fraction of  $\text{H}_2\text{O}$ . If BURNER = T (see Card 20) and a water-producing fuel is used, XH2 is added to the resulting water fraction.

XCO - mole fraction of He.

JJ - rotational quantum number at which gain is computed.

NOTE: Gain is also calculated at the rotational quantum number at which it is greatest.

Card 31: ITAUSB, IRSTRT, IFLUX, ISUB, ITABLE, IRATE, IH2, IJ, IPRINT (911)

ITAUSB - water rate flag.

0: use nominal rate.

1: use Sharma rate.

IRSTRT - restart flag.

- 0: don't use restart option.
- 1: use restart option.

IFLUX - flux flag.

- 0: cavity flux is not present.
- 1: cavity flux is present.

ISUB - subsonic nozzle contour flag.

- 0: use stored values.
- 1: use external contour (see Cards 40-41)
- 2: use CONTOUR subroutine (see Card 42)
- 3: use previous values.
- 4: special nozzle 1.
- 5: special nozzle 2.
- 6: special nozzle 3.
- 7: use Mach number derivative subroutine.

ITABLE - supersonic nozzle contour flag.

- 0: use default contour (method-of-characteristics contour for a Laval nozzle)
- 1: use table look-up.

IRATE - vibrational probabilities flag.

- 1: use external values (see Cards 34-39).
- 0: use stored values.

IR2 - hydrogen rates flag.

- 1: don't use AFWL hydrogen rates.
- 2: use AFWL hydrogen rates.

IJ - inactive plot parameter. It should be set to 0.

IPRINT - print flag.

- 0: normal print.
- 1: additional print at every step from  $X = 0.05$  cm to  $X = 0.5$  cm in the cavity.

Card 32: DELX, ELSUB, ELNOZ, SLOPE, CURVE (5F10.0)

DELX - print interval for cavity calculations, in cm.

ELSUB - length of converging (subsonic) portion of the nozzle, in cm.  $A/A^* = 10$  at  $X = \text{ELSUB}$ .

ELNOZ - length of diverging (supersonic) portion of the nozzle, in cm.

SLOPE - slope of the nozzle contour at  $X = 0$ . It is ignored if  $\text{ELNOZ} = 0$ .

CURVE - radius of curvature of the nozzle contour at  $X = 0$ .

Card 33: PHI, XBMON, XBMOFF (3F10.0)

PHI - cavity optical flux, in watts/cm<sup>2</sup>.

XBMON - downstream distance where flux starts, in cm.

XBMOFF - downstream distance where flux stops, in cm.

NOTE: Include this card only if  $\text{IFLUX} \neq 0$  (see Card 31).

Card 34: B1, B2, B3, B4 (4F18.0)

Card 35: B5, B6, B7, B8 (4F18.0)

Card 37: B19 (F18.0)

B1-B19 - coefficients of probability functions describing vibrational relaxation processes. They are specified in the form  $B_i F_i(T)$ ; where  $B_i$  are constants and  $F_i$  are specific polynomials contained in subroutine TAUSUB.

NOTE: Include these cards only if  $\text{IRATE} = 1$  (see Card 31).

Card 38: TAU31, SIGC, SIGN, SIGN2 (4F18.0)

Card 39: SIGHE (F18.0)

TAU31 - transitional lifetime for the  $\text{CO}_2(001) \rightarrow \text{CO}_2(100)$  transition, in seconds.

SIGC - optical broadening collision cross section for collisions between  $\text{CO}_2$ , in cm<sup>2</sup>.

SIGN - optical broadening collision cross section for collisions between  $\text{N}_2$ , in cm<sup>2</sup>.



SIGH2 - optical broadening collision cross section for collisions between  $H_2O$ , in  $cm^2$ .

SIGHE - optical broadening collision cross section for collisions between He, in  $cm^2$ .

NOTE: Include these cards only if IRATE = 1 (see Card 31).

Card 40: CM1, CM2, CM3, CM4 (4F18.0)

Card 41: CM5 (F18.0)

CM1-CM5 - coefficients for polynomial fit to subsonic nozzle contour.

NOTE: Include these cards only if ISUB = 1 (see Card 31).

Card 42: C, R, THETA1, THETA2 (4F10.0)

C - length, perpendicular to the throat plane, of the first section of the subsonic portion of the nozzle, in cm.

R - radius of the circular arc defining the second section of the nozzle, in cm.

THETA1 - angle orienting the first nozzle section in degrees. The first section of the nozzle is at an angle of (90-THETA1) to the nozzle throat plane.

THETA2 - angle orienting the second nozzle section in degrees. This section is a straight line, tangent to the circle at an angle of (90-THETA2) to the nozzle throat plane.

NOTE: Include these cards only if ISUB = 2 (see Card 31).

Card 43: N, X, EM, T (4F18.0)

Card 44: TVC1, TVC2, TVC3, TVN (4F18.0)

Card 45: TVH2, TVHH, P, A (4F18.0)

N - mach step value at which calculation is to be restarted.

X - X coordinate at which calculation is to be restarted (corresponds to N), in cm.

EM - Mach number at X.

T - gas temperature at X, in degrees Kelvin.

TVC1 - vibrational temperature at X of mode  $v_1$  of  $\text{CO}_2$ , in degrees Kelvin.

TVC2 - vibrational temperature at X of mode  $v_2$  of  $\text{CO}_2$ , in degrees Kelvin.

TVC3 - vibrational temperature at X of mode  $v_3$  of  $\text{CO}_2$ , in degrees Kelvin.

TVN - vibrational temperature at X of  $\text{N}_2$ , in degrees Kelvin.

TVH2 - vibrational temperature at X of mode  $v_2$  of  $\text{H}_2\text{O}$ , in degrees Kelvin.

TVHH - vibrational temperature at X of  $\text{H}_2$ , in degrees Kelvin.

P - static pressure at X, in atmospheres.

A - nozzle to throat area ratio at X.

NOTE: Include this card only ifIRSTRT  $\neq$  0 (see Card 31).

This completes the input. Additional problems can be run by repeating Cards 1-41. To terminate, a card with 'STOP' is required.

#### SAMPLE CALCULATION

This section presents the results of a typical optimization problem. It utilizes virtually all features of the model except that weight and volume results are not used. This calculation maximizes the available specific energy of a  $\text{CO}/\text{N}_2\text{O}$  fuel/oxidizer device constrained to recover to atmospheric pressure. The design variables, constraints, and objective function are summarized in table 3. These choices are based on reference 2 and represent a reasonably advanced laser.

TABLE 3.- OPTIMIZATION PARAMETERS

#### Design variables

Variable	Initial value	Lower bound	Upper bound
Stagnation pressure, atm	60	25	115
Nozzle throat height, mm	.20	0.15	10
Nozzle area ratio	50	10	80
Mole fraction of $\text{H}_2\text{O}$	.005	.001	.05
Moles of $\text{N}_2$ to moles of C-N-O-H	8	3.5	15

TABLE 3.- Concluded

Constraints			
<u>Constraint</u>	<u>Initial value</u>	<u>Lower bound</u>	<u>Upper bound</u>
Stagnation temperature, K	1338	1000	2100
Peak gain, %cm <sup>-1</sup>	1.32	1	5
Actual minus required pressure at end of diffuser ( $\geq 0$ implies recovery)	314.5	0	1000
<u>Objective function</u>	<u>Initial value</u>	<u>Final value</u>	
Maximum available specific energy	9.92 kJ/lb	16.00 kJ/lb	

Shown below are the input data (fig. 2) and the resulting output (fig. 3) for this problem.

ORIGINAL PAGE IS  
OF POOR QUALITY

# GDL OPTIMIZATION SAMPLE, JULY 1977

2	5	5	1	
5	05	3	3	3
0.02	0.01			0
5	16	1.0		
25.0	115.0			
0.15	10.0			
10.0	80.0			
0.001	0.05			
3.5	15.0			
1	2	1.0		
2	3	1.0		
3	4	1.0		
4	6	1.0		
5	19	1.0		
3				
1	0	1000.	2100.0	1000.
15	0	1.0	5.0	1.0
34	0	0.0	1000.0	1000.0

END

\$OPTION BURNER=T, WEIGHT=T, VLCAV=F, LASER=T, \$END

\$SIZE ETA=0.8, EMEXIT=0.2, \$END

\$PROPER CHI=0., Q=0., R=8., S=0., KFUEL=1, KOXID=2, \$END

1

2OPTIMUM GDL

N2O OXIDIZER

3

0

GDL OPTIMIZATION SAMPLE

1700.0	60.0	50.0	0.200	50.0	0.0
--------	------	------	-------	------	-----

0.13	0.005	0.0	0.0		
------	-------	-----	-----	--	--

000600200

2.0	0.1628	4.37	0.	0.
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STOP

\*\*

Figure 2.- Data input.



ORIGINAL PAGE IS  
OF POOR QUALITY

TITLE:  
GOL OPTIMIZATION SAMPLE, JULY 1977

CONTROL PARAMETERS:  
CALCULATION CONTROL, NCALL = 2  
NUMBER OF GLOBAL DESIGN VARIABLES, NSV = 5  
NUMBER OF SENSITIVITY VARIABLES, NSV = 0  
NUMBER OF FUNCTIONS IN TWO-SPACE, N2VAR = 0  
INPUT INFORMATION PRINT CODE, IPINPT = 0  
SENSITIVITY PRINT CODE, IPSENS = 1  
TWO-SPACE PRINT CODE, IP2VAR = 0  
DEBUG PRINT CODE, IPDBG = 0

CALCULATION CONTROL, NCALL  
VALUE MEANING  
1 SINGLE ANALYSIS  
2 OPTIMIZATION  
3 SENSITIVITY  
4 OPTIMUM SENSITIVITY  
5 TWO-VARIABLE FUNCTION SPACE  
6 OPTIMUM TWO-VARIABLE FUNCTION SPACE

#### OPTIMIZATION INFORMATION

GLOBAL VARIABLE NUMBER 16 WILL BE MAXIMIZED  
DESIGN VARIABLE INFORMATION  
NON-ZERO INITIAL VALUE WILL GIVE-RISE MODULE INPUT  
D. V. LOWER UPPER INITIAL VALUE  
NO. BOUND  
1 .2500E+02 .1150E+03 0.  
2 .1500E+00 .1000E+02 0.  
3 .1000E+02 .5000E+02 0.  
4 .1000E-02 .5000E-01 0.  
5 .3500E+01 .1500E+02 0.

DESIGN VARIABLES GLOBAL MULTIPLYING  
D. V. GLOBAL FACTOR  
ID NO. VAL. NO.  
1 1 2  
2 2 3  
3 3 4  
4 4 6  
5 5 19

#### CONSTRAINT INFORMATION

THERE ARE 3 CONSTRAINT SETS

IN GLOBAL GLOBAL LINEA

VAL. 1 VAL. 2

1 1 0 0

2 15 0 0

3 3 0 0

4 19 0 0

TOTAL NUMBER OF CONSTRAINT PARAMETERS = 3

UPPER  
BOUND  
21000E+04  
5000E+01  
1000E+02

LOWER  
BOUND  
1000E+04  
1000E+01  
1000E+02



## \*\*\*\*\*FLAME RESULTS\*\*\*\*\*

THE FUEL IS CO  
THE OXIDIZER IS N2 O

CHI = 0.0000 Q = 0.0000 R = 2.0000 S = 0.0000 ET4N2 = 1.0000

\*\*\*\*\*REACTANTS\*\*\*\*\*  
MASS MOLE  
CARBON 0.0000 0.0000  
CHON .0946 .1000  
N1 .1446 .1000  
N2 .7500 .5000  
CO2 0.0000 0.0000

\*\*\*\*\*PRODUCTS\*\*\*\*\*  
MASS MOLE  
CO2 .1446 .1000  
H2O 0.0000 0.0000  
O2 0.0000 0.0000  
N2 .6514 .5000

FUEL-AIR RATIO = .1045 EXCESS AIR = 0.0000

FLAME TEMP = 1337.2986 DEG. K  
DELTX = 0.000 DEG. K  
DELN2 = 0.000 DEG. K

ADDITIONAL N2O ENTHALPY = 0.  
ADDITIONAL ENTHALPY = 0.

BTU/MOLE  
BTU/MOLE FOR 1.00 OF THE TOTAL NITROGEN

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*****AVCO G-IN MODEL* NUCLEARMS VERSION 2.0*****
      GOL OPTIMIZE: ION SAMPLE
      OPTIMUM GOL
      NUCOXADIZE

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NOZZLE HEIGHT (MM) = .2000  
NOMINAL RATE FOR PC2H

<hr/>					
TSRAG = 1317.99 DEG K					
C02 FRACTION = .10000	H2O FRACTION = .00500	FSTAG = 60.0000 ATM	LEWIT/M = 50.00	XMAX = 30.00	ANGLE = 0.00
N2 FRACTION = .89500	J = 0	ITAUSP = 0	CO FRACTION = .00000	HE FRACTION = 0.00000	M2 FRACTION = .00000
IPLOT = 3	IJ = 0	DELX = 2.0000	ELSUB = .103957 CM	ELADZ = 4.87000 LM	SLOPE = 23.356793
CURVE = 0.00000	KMAX = 100000	DYMAX = .500000	ZETA1 = 0.0000	TMETH2 = 0.0000	FROZEN GAMMA = 1.39960040
C = 0.000000	LAL = 0.000000	CM2 = .697939E+02	CMX = .126143E+03	CM4 = -.764232E+02	CM5 = -.351523E+03
CH1 = .131456E+02	THE STANDARD DEVIATION FOR THE LMS IS 0.000000	THE NUMBER OF POINTS TAKEN IS 0			
FRICION FACTOR = 0.00000	PLOCEID GAMMA = 1.20512	WEIGHTED GAS CONSTANT OF MIXTURE EGAS/GM-DEG K = .29122E+07			
SWITCHOVER AT M = 1.05000	Z/LA = 1.002026	X = .00002 CM			
B1 = .100E+01	B2 = .500E+00	B3 = .100E+01	B4 = .143E+00	B5 = .100E+01	B6 = .100E+01
B7 = .100E+01	B8 = .10E+01	B9 = .100E+01	B10 = .100E+01	B17 = .100E+01	B18 = .100E+01
B19 = .100E+01					
TRANSITION LIFETIME = .536E+01 SEC					
OPTICAL CROSS SECTION = .170E+19 CM2 PER MOL.					
= .370E+14 CM2 PER MOL.					
= .300E+14 CM2 PER MOL.					
= .370E+14 CM2 PER MOL.					



[illegible]

1/LN1	1/LN2	1/LN3	1/LN4	1/LN5	1/LN6	1/LN7	1/LN8	1/LN9	1/LN10	1/LN11	1/LN12	1/LN13	1/LN14	1/LN15	1/LN16	1/LN17	1/LN18	1/LN19	1/LN20	1/LN21	1/LN22	1/LN23	1/LN24	1/LN25	1/LN26	1/LN27	1/LN28	1/LN29	1/LN30	1/LN31	1/LN32	1/LN33	1/LN34	1/LN35	1/LN36	1/LN37	1/LN38	1/LN39	1/LN40	1/LN41	1/LN42	1/LN43	1/LN44	1/LN45	1/LN46	1/LN47	1/LN48	1/LN49	1/LN50	1/LN51	1/LN52	1/LN53	1/LN54	1/LN55	1/LN56	1/LN57	1/LN58	1/LN59	1/LN60	1/LN61	1/LN62	1/LN63	1/LN64	1/LN65	1/LN66	1/LN67	1/LN68	1/LN69	1/LN70	1/LN71	1/LN72	1/LN73	1/LN74	1/LN75	1/LN76	1/LN77	1/LN78	1/LN79	1/LN80	1/LN81	1/LN82	1/LN83	1/LN84	1/LN85	1/LN86	1/LN87	1/LN88	1/LN89	1/LN90	1/LN91	1/LN92	1/LN93	1/LN94	1/LN95	1/LN96	1/LN97	1/LN98	1/LN99	1/LN100	1/LN101	1/LN102	1/LN103	1/LN104	1/LN105	1/LN106	1/LN107	1/LN108	1/LN109	1/LN110	1/LN111	1/LN112	1/LN113	1/LN114	1/LN115	1/LN116	1/LN117	1/LN118	1/LN119	1/LN120	1/LN121	1/LN122	1/LN123	1/LN124	1/LN125	1/LN126	1/LN127	1/LN128	1/LN129	1/LN130	1/LN131	1/LN132	1/LN133	1/LN134	1/LN135	1/LN136	1/LN137	1/LN138	1/LN139	1/LN140	1/LN141	1/LN142	1/LN143	1/LN144	1/LN145	1/LN146	1/LN147	1/LN148	1/LN149	1/LN150	1/LN151	1/LN152	1/LN153	1/LN154	1/LN155	1/LN156	1/LN157	1/LN158	1/LN159	1/LN160	1/LN161	1/LN162	1/LN163	1/LN164	1/LN165	1/LN166	1/LN167	1/LN168	1/LN169	1/LN170	1/LN171	1/LN172	1/LN173	1/LN174	1/LN175	1/LN176	1/LN177	1/LN178	1/LN179	1/LN180	1/LN181	1/LN182	1/LN183	1/LN184	1/LN185	1/LN186	1/LN187	1/LN188	1/LN189	1/LN190	1/LN191	1/LN192	1/LN193	1/LN194	1/LN195	1/LN196	1/LN197	1/LN198	1/LN199	1/LN200	1/LN201	1/LN202	1/LN203	1/LN204	1/LN205	1/LN206	1/LN207	1/LN208	1/LN209	1/LN210	1/LN211	1/LN212	1/LN213	1/LN214	1/LN215	1/LN216	1/LN217	1/LN218	1/LN219	1/LN220	1/LN221	1/LN222	1/LN223	1/LN224	1/LN225	1/LN226	1/LN227	1/LN228	1/LN229	1/LN230	1/LN231	1/LN232	1/LN233	1/LN234	1/LN235	1/LN236	1/LN237	1/LN238	1/LN239	1/LN240	1/LN241	1/LN242	1/LN243	1/LN244	1/LN245	1/LN246	1/LN247	1/LN248	1/LN249	1/LN250	1/LN251	1/LN252	1/LN253	1/LN254	1/LN255	1/LN256	1/LN257	1/LN258	1/LN259	1/LN260	1/LN261	1/LN262	1/LN263	1/LN264	1/LN265	1/LN266	1/LN267	1/LN268	1/LN269	1/LN270	1/LN271	1/LN272	1/LN273	1/LN274	1/LN275	1/LN276	1/LN277	1/LN278	1/LN279	1/LN280	1/LN281	1/LN282	1/LN283	1/LN284	1/LN285	1/LN286	1/LN287	1/LN288	1/LN289	1/LN290	1/LN291	1/LN292	1/LN293	1/LN294	1/LN295	1/LN296	1/LN297	1/LN298	1/LN299	1/LN300	1/LN301	1/LN302	1/LN303	1/LN304	1/LN305	1/LN306	1/LN307	1/LN308	1/LN309	1/LN310	1/LN311	1/LN312	1/LN313	1/LN314	1/LN315	1/LN316	1/LN317	1/LN318	1/LN319	1/LN320	1/LN321	1/LN322	1/LN323	1/LN324	1/LN325	1/LN326	1/LN327	1/LN328	1/LN329	1/LN330	1/LN331	1/LN332	1/LN333	1/LN334	1/LN335	1/LN336	1/LN337	1/LN338	1/LN339	1/LN340	1/LN341	1/LN342	1/LN343	1/LN344	1/LN345	1/LN346	1/LN347	1/LN348	1/LN349	1/LN350	1/LN351	1/LN352	1/LN353	1/LN354	1/LN355	1/LN356	1/LN357	1/LN358	1/LN359	1/LN360	1/LN361	1/LN362	1/LN363	1/LN364	1/LN365	1/LN366	1/LN367	1/LN368	1/LN369	1/LN370	1/LN371	1/LN372	1/LN373	1/LN374	1/LN375	1/LN376	1/LN377	1/LN378	1/LN379	1/LN380	1/LN381	1/LN382	1/LN383	1/LN384	1/LN385	1/LN386	1/LN387	1/LN388	1/LN389	1/LN390	1/LN391	1/LN392	1/LN393	1/LN394	1/LN395	1/LN396	1/LN397	1/LN398	1/LN399	1/LN400	1/LN401	1/LN402	1/LN403	1/LN404	1/LN405	1/LN406	1/LN407	1/LN408	1/LN409	1/LN410	1/LN411	1/LN412	1/LN413	1/LN414	1/LN415	1/LN416	1/LN417	1/LN418	1/LN419	1/LN420	1/LN421	1/LN422	1/LN423	1/LN424	1/LN425	1/LN426	1/LN427	1/LN428	1/LN429	1/LN430	1/LN431	1/LN432	1/LN433	1/LN434	1/LN435	1/LN436	1/LN437	1/LN438	1/LN439	1/LN440	1/LN441	1/LN442	1/LN443	1/LN444	1/LN445	1/LN446	1/LN447	1/LN448	1/LN449	1/LN450	1/LN451	1/LN452	1/LN453	1/LN454	1/LN455	1/LN456	1/LN457	1/LN458	1/LN459	1/LN460	1/LN461	1/LN462	1/LN463	1/LN464	1/LN465	1/LN466	1/LN467	1/LN468	1/LN469	1/LN470	1/LN471	1/LN472	1/LN473	1/LN474	1/LN475	1/LN476	1/LN477	1/LN478	1/LN479	1/LN480	1/LN481	1/LN482	1/LN483	1/LN484	1/LN485	1/LN486	1/LN487	1/LN488	1/LN489	1/LN490	1/LN491	1/LN492	1/LN493	1/LN494	1/LN495	1/LN496	1/LN497	1/LN498	1/LN499	1/LN500	1/LN501	1/LN502	1/LN503	1/LN504	1/LN505	1/LN506	1/LN507	1/LN508	1/LN509	1/LN510	1/LN511	1/LN512	1/LN513	1/LN514	1/LN515	1/LN516	1/LN517	1/LN518	1/LN519	1/LN520	1/LN521	1/LN522	1/LN523	1/LN524	1/LN525	1/LN526	1/LN527	1/LN528	1/LN529	1/LN530	1/LN531	1/LN532	1/LN533	1/LN534	1/LN535	1/LN536	1/LN537	1/LN538	1/LN539	1/LN540	1/LN541	1/LN542	1/LN543	1/LN544	1/LN545	1/LN546	1/LN547	1/LN548	1/LN549	1/LN550	1/LN551	1/LN552	1/LN553	1/LN554	1/LN555	1/LN556	1/LN557	1/LN558	1/LN559	1/LN560	1/LN561	1/LN562	1/LN563	1/LN564	1/LN565	1/LN566	1/LN567	1/LN568	1/LN569	1/LN570	1/LN571	1/LN572	1/LN573	1/LN574	1/LN575	1/LN576	1/LN577	1/LN578	1/LN579	1/LN580	1/LN581	1/LN582	1/LN583	1/LN584	1/LN585	1/LN586	1/LN587	1/LN588	1/LN589	1/LN590	1/LN591	1/LN592	1/LN593	1/LN594	1/LN595	1/LN596	1/LN597	1/LN598	1/LN599	1/LN600	1/LN601	1/LN602	1/LN603	1/LN604	1/LN605	1/LN606	1/LN607	1/LN608	1/LN609	1/LN610	1/LN611	1/LN612	1/LN613	1/LN614	1/LN615	1/LN616	1/LN617	1/LN618	1/LN619	1/LN620	1/LN621	1/LN622	1/LN623	1/LN624	1/LN625	1/LN626	1/LN627	1/LN628	1/LN629	1/LN630	1/LN631	1/LN632	1/LN633	1/LN634	1/LN635	1/LN636	1/LN637	1/LN638	1/LN639	1/LN640	1/LN641	1/LN642	1/LN643	1/LN644	1/LN645	1/LN646	1/LN647	1/LN648	1/LN649	1/LN650	1/LN651	1/LN652	1/LN653	1/LN654	1/LN655	1/LN656	1/LN657	1/LN658	1/LN659	1/LN660	1/LN661	1/LN662	1/LN663	1/LN664	1/LN665	1/LN666	1/LN667	1/LN668	1/LN669	1/LN670	1/LN671	1/LN672	1/LN673	1/LN674	1/LN675	1/LN676	1/LN677	1/LN678	1/LN679	1/LN680	1/LN681	1/LN682	1/LN683	1/LN684	1/LN685	1/LN686	1/LN687	1/LN688	1/LN689	1/LN690	1/LN691	1/LN692	1/LN693	1/LN694	1/LN695	1/LN696	1/LN697	1/LN698	1/LN699	1/LN700	1/LN701	1/LN702	1/LN703	1/LN704	1/LN705	1/LN706	1/LN707	1/LN708	1/LN709	1/LN710	1/LN711	1/LN712	1/LN713	1/LN714	1/LN715	1/LN716	1/LN717	1/LN718	1/LN719	1/LN720	1/LN721	1/LN722	1/LN723	1/LN724	1/LN725	1/LN726	1/LN727	1/LN728	1/LN729	1/LN730	1/LN731	1/LN732	1/LN733	1/LN734	1/LN735	1/LN736	1/LN737	1/LN738	1/LN739	1/LN740	1/LN741	1/LN742	1/LN743	1/LN744	1/LN745	1/LN746	1/LN747	1/LN748	1/LN749	1/LN750	1/LN751	1/LN752	1/LN753	1/LN754	1/LN755	1/LN756	1/LN757	1/LN758	1/LN759	1/LN760	1/LN761	1/LN762	1/LN763	1/LN764	1/LN765	1/LN766	1/LN767	1/LN768	1/LN769	1/LN770	1/LN771	1/LN772	1/LN773	1/LN774	1/LN775	1/LN776	1/LN777	1/LN778	1/LN779	1/LN780	1/LN781	1/LN782	1/LN783	1/LN784	1/LN785	1/LN786	1/LN787	1/LN788	1/LN789	1/LN790	1/LN791	1/LN792	1/LN793	1/LN794	1/LN795	1/LN796	1/LN797	1/LN798	1/LN799	1/LN800	1/LN801	1/LN802	1/LN803	1/LN804	1/LN805	1/LN806	1/LN807	1/LN808	1/LN809	1/LN810	1/LN811	1/LN812	1/LN813	1/LN814	1/LN815	1/LN816	1/LN817	1/LN818	1/LN819	1/LN820	1/LN821	1/LN822	1/LN823	1/LN824	1/LN825	1/LN826	1/LN827	1/LN828	1/LN829	1/LN830	1/LN831	1/LN832	1/LN833	1/LN834	1/LN835	1/LN836	1/LN837	1/LN838	1/LN839	1/LN840	1/LN841	1/LN842	1/LN843	1/LN844	1/LN845	1/LN846	1/LN847	1/LN848	1/LN849	1/LN850	1/LN851	1/LN852	1/LN853	1/LN854	1/LN855	1/LN856	1/LN857	1/LN858	1/LN859	1/LN860	1/LN861	1/LN862	1/LN863	1/LN864	1/LN865	1/LN866	1/LN867	1/LN868	1/LN869	1/LN870	1/LN871	1/LN872	1/LN873	1/LN874	1/LN875	1/LN876	1/LN877	1/LN878	1/LN879	1/LN880	1/LN881	1/LN882	1/LN883	1/LN884	1/LN885	1/LN886	1/LN887	1/LN888	1/LN889	1/LN890	1/LN891	1/LN892	1/LN893	1/LN894	1/LN895	1/LN896	1/LN897	1/LN898	1/LN899	1/LN900	1/LN901	1/LN902	1/LN903	1/LN904	1/LN905	1/LN906	1/LN907	1/LN908	1/LN909	1/LN910	1/LN911	1/LN912	1/LN913	1/LN914	1/LN915	1/LN916	1/LN917	1/LN918	1/LN919	1/LN920	1/LN921	1/LN922	1/LN923	1/LN924	1/LN925	1/LN926	1/LN927	1/LN928	1/LN929	1/LN930	1/LN931	1/LN932	1/LN933	1/LN934	1/LN935	1/LN936	1/LN937	1/LN938	1/LN939	1/LN940	1/LN941	1/LN942	1/LN943	1/LN944	1/LN945	1/LN946	1/LN947	1/LN948	1/LN949	1/LN950	1/LN951	1/LN952	1/LN953	1/LN954	1/LN955	1/LN956	1/LN957	1/LN958	1/LN959	1/LN960	1/LN961	1/LN962	1/LN963	1/LN964	1/LN965	1/LN966	1/LN967	1/LN968	1/LN969	1/LN970	1/LN971	1/LN972	1/LN973	1/LN974	1/LN975	1/LN976	1/LN977	1/LN978	1/LN979	1/LN980	1/LN981	1/LN982	1/LN983	1/LN984	1/LN985	1/LN986	1/LN987	1/LN988	1/LN989	1/LN990	1/LN991	1/LN992	1/LN993	1/LN994	1/LN995	1/LN996	1/LN997	1/LN998	1/LN999	1/LN1000
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[illegible]



ORIGINAL PAGE IS  
OF POOR QUALITY

X (CM)	M	A/A	P (AIMOS)	T (DEG K)	T01 (DEG K)	T-2 (DEG K)	T-3 (DEG K)	T-4 (DEG K)	T-5 (DEG K)
DX	DM/EX	D/DX (A/A)	EP/EX	DT/EX	DT01/EX	DT02/EX	DT03/EX	DT04/EX	DT05/EX
1/LNC	U (CM/SEC)	EC1/R (DEG K)	EC2/R (DEG K)	EC3/R (DEG K)	EN/R (DEG K)	EH2/R (DEG K)	EH3/R (DEG K)	EH4/R (DEG K)	EH5/R (DEG K)
1/LNHE	1/LC2N	1/LC2H	1/LC2H	1/LC3H	1/LNHE	1/LC3H	1/LC3H	1/LC3H	1/LC3H
GAINJP (1/CM)	1/LC32HE	HST/R (DEG K)	T-20 (DEG K)	EC0/R (DEG K)	DT00/EX	TH2 (DEG K)	EH2/R (DEG K)	EH3/R (DEG K)	EH4/R (DEG K)
	GAINJR (1/CM)	FLJPR (W/CM2)	MAX GAIN	J/LB -3MAX	J AT GMAX	PHI (W/CM2)	DCP F40 P	JUP F40 R	DENSE (G/CM3)

\*\*\*\*\* OPTIMUM GOL  
\*\*\*\*\* M20 OXIDIZER

#### GOL OPTIMIZATION SAMPLE

THE MAXIMUM GAIN IS  $1.31E+01$  DC/CM THIS OCCURS AT  $V = 6.06371$  UM  
THERE IS NO VALUE OF GAIN SUITABLE FOR THE REQUIRED CONDITION  
THE NUMBER CHARACTERIZING THE TRAILING PORTION OF THE CURVE IS  $3.124671$   
THE NUMBER CHARACTERIZING THE CURVATURE OF THE TRAILING PORTION OF THE CURVE IS  $.4547424$   
THE NUMBER AVERAGE SLOPE IS  $-.00953$

THE AVERAGE SPECIFIC POWER OVER A 10.000 CM DISTANCE IS  $351.2470$  J/LB.  
IT WAS AVERAGED BETWEEN 2.016 CM AND 12.016 CM.

## \*\*\*WEIGHT/VOLUME RESULTS\*\*\*

TOTAL PRESSURE (PSI)	862.00	IGNITION TEMPERATURE (DEGREE CELSIUS)	1337.39
TOTAL POWER (WATTS)	500000.00	SPECIFIC POWER (JOULES/GRAM) (EFF = 1.00)	9913.25
NUMBER OF SHOTS	15	SHOT LENGTH (SECONDS/FIRING)	2.00
TOTAL FLOW TIME (SECONDS)	40.00	MASS FLOW RATE (LBM/SECOND)	50.41
LASER AND MIRROR HEIGHT (INCHES)	5.00	LASER WIDTH (INCHES)	71.07
CAVITY ASPECT RATIO	14.21	NOZZLE BOUNDARY LAYER THICKNESS	.965417E-05
DIFFUSER LENGTH (INCHES)	6.46	REQUIRED DIFFUSER ENTRANCE PRESSURE (PSI)	523.30
DIFFUSER EXIT MACH NUMBER	.20	DIFFUSER EXIT PRESSURE (PSI)	14.70

## SYSTEM AND SUBSYSTEM WEIGHTS AND VOLUMES

	WEIGHT (LB)	VOLUME (F <sup>3</sup> )*3
• COMBUSTION	120.46	3.96%
• CAVITY, NOZZLE, MANIFOLD	413.95	5.11%
SIDEWALL	112.05	
NOZZLE	224.11	
ENDWALL	60.00	
MANIFOLD	17.79	
• STRUCTURE	103.29	
• MOUNTS	41.50	
• DIFFUSER (ET) = .10	462.72	1.33%
WIND (55)	133.90	
SIDEWALL	27.03	
ENDWALL	320.54	
MOUNTS	11.25	
• OPTICAL DENS	36.07	9.92%
• ADVANCED OPTICAL BENCH	500.00	10.00%
• FUEL SUPPLY SYSTEM	2376.41	120.15%
FUEL TANK	2037.46	41.74%
PUMP	62.62	
VALVE	126.02	4.03%
MISCELLANEOUS	100.31	33.31%
TOTAL	351.00	16.00%
TOTAL USING ADVANCED BENCH	3936.10	154.56%
	4400.03	156.63%

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# C O N M I N

F O R T R A N P R O G R A M F O R

C O N S T R A I N E D F U N C T I O N M I N I M I Z A T I O N

N A S A J A M E S R E S E A R C H C E N T E R , M O F F E T T F I E L D , C A L I F .

V E R S I O N 11 J U L Y , 1 9 7 5

## C O N S T R A I N E D F U N C T I O N M I N I M I Z A T I O N

### C O N T R O L P A R A M E T E R S

IPRINT	NDV	ITMAX	NCN	NSIDE	LCNDR	NSCAL	WFDS
5	5	5	6	1	3	3	0
LINOBJ	ITPM	N1	N2	N3	N4	N5	
0	3	7	16	7	7	14	
CT	CTMIN	CIL	CTLMN				
- .5000E-01	.4000E-02	- .1000E-01	.1000E-02				
THETA	PMT	DELFUN	DABFUN				
.1000E+01	.5000E+01	.1000E-02	.95152E+01				
FDCH	FDCHM						
.2000E-01	.1000E-01						

LOWER BOUNDS ON DECISION VARIABLES (VLB)

1)	.2500E+02	.1500E+00	.1000E+02	.1000E-02	.3500E+01
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UPPER BOUNDS ON DECISION VARIABLES (VUB)

1)	.1150E+03	.1000E+02	.8000E+02	.5000E-01	.1500E+02
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ALL CONSTRAINTS ARE NON-LINEAR

### I N I T I A L F U N C T I O N I N F O R M A T I O N

OBJ = -.391925E+04

### D E C I S I O N V A R I A B L E S (X-VECTOR)

1)	.6000E+02	.2000E+00	.5000E+02	.5000E-02	.3000E+01
----	-----------	-----------	-----------	-----------	-----------

### C O N S T R A I N T V A L U E S (G-VECTOR)

1)	-.31789E+00	-.76211E+00	-.31888E+00	-.36311E+01	-.25670E+02	.15170E+02
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BEGIN LINE-ATION NUMBER 1
CT = -.50000E-01 CYL = -.10000E-01 PHZ = .50000E+01
NEW SCALING VECTOR (SCAL)
.6000E+02 .2000E+00 .5000E+02 .5000E-02 .2000E+01
THERE ARE 0 ACTIVE CONSTRAINTS
THERE ARE 1 VIOLATED CONSTRAINTS
CONSTRAINT NUMBERS ARE
6
THERE ARE 0 ACTIVE SIDE CONSTRAINTS
GRADIENT OF OBJ
1) .21294E+04 .74930E+03 -.11810E+04 .52204E+03 .10786E+05
GRADIENTS OF ACTIVE AND VIOLATED CONSTRAINTS
CONSTRAINT NUMBER 6
1) .82709E+02 -.35049E+01 -.66339E+02 -.5165E-01 -.43752E+02
PUSH-OFF FACTORS, (THETA(1), T=1.0AC)
1) .10000E+01
CONSTRAINT PARAMETER, BETA = .11789E+01
SEARCH DIRECTION (S-VECTOR)
1) -.10000E+01 .55741E-02 .73774E+00 -.2231E-01 .68019E-02
ONE-DIMENSIONAL SEARCH
INITIAL SLOPE = -.3151E+04 PROPOSED ALPHA = .1173E+00

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• • CONSTRAINED ONE-DIMENSIONAL SEARCH INFORMATION • •

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PROPOSED DESIGN
ALPHA = .11731E+00
X-VECTOR
.5296E+02 .2001E+00 .5462E+02 .4987E-02 .8006E+01
OBJ = -.10314E+05
CONSTRAINT VALUES
-.3373E+00 -.7627E+00 -.3648E+00 -.3635E+01 -.1036E+02 .3645E+00
TWO-POINT INTERPOLATION
PROPOSED DESIGN
ALPHA = .12007E+00
X-VECTOR
.5280E+02 .2001E+00 .5473E+02 .4337E-02 .3007E+01
OBJ = -.10324E+05
CONSTRAINT VALUES
-.3373E+00 -.7627E+00 -.3666E+00 -.3633E+01 -.1000E+02 .3639E-02

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• • • END OF ONE-DIMENSIONAL SEARCH

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CALCULATED ALPHA = .1200/E+00
OBJ = -.103240E+05
DECISION VARIABLES (X-VECTOR)
1) .52796E+02 .20013E+00 .54729E+02 .49866E-02 .00665E+01
CONSTRAINT VALUES (G-VECTOR)
1) -.33729E+00 -.76271E+00 -.36659E+00 -.36334E+01 -.10004E+02 .36392E-02

BEGIN ITERATION NUMBER 2
CT = -.50000E-01 CYL = -.10000E-01 FMI = .50000E+01
THERE ARE 1 ACTIVE CONSTRAINTS
CONSTRAINT NUMBERS ARE
5
THERE ARE 0 VIOLATED CONSTRAINTS
THERE ARE 0 ACTIVE SIDE CONSTRAINTS
GRADIENT OF OBJ
1) .24555E+04 .72155E+03 -.12634E+04 .43266E+03 .11853E+05
GRADIENTS OF ACTIVE AND VIOLATED CONSTRAINTS
CONSTRAINT NUMBER 5
1) .78156E+02 -.43545E+01 -.63375E+02 -.14110E+01 -.50233E+02

PUSH-OFF FACTORS, (THETA(I), I=1,NAC)
1) .11509E+01
CONSTRAINT PARAMETER, BETA = .77271E+00
SEARCH DIRECTION (S-VECTOR)
1) -.10000E+01 -.17716E-01 .74912E+00 -.82751E-01 -.50613E+00
ONE-DIMENSIONAL SEARCH
INITIAL SLOPE = -.9431E+04 PROPOSED ALPHA = .5474E-01

... CONSTRAINED ONE-DIMENSIONAL SEARCH INFORMATION ...

PROPOSED DESIGN
ALPHA = .54736E-01
X-VECTOR
.4951E+02 .1936E+00 .5673E+02 .4380E-02 .7785E+01
OBJ = -.10847E+05
CONSTRAINT VALUES
-.3582E+00 -.7416E+00 -.4122E+00 -.3584E+01 -.4747E+01 -.5253E+01
TWO-POINT INTERPOLATION
PROPOSED DESIGN
ALPHA = .10416E+00

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X-VECTOR      .1988E+00   .583E+02   .4975E-02   .7565E+01
OBJ = -.11319E+05
CONSTRAINT VALUES
-.1779E+00   -.7221E+00   -.4457E+00   -.3554E+01   -.3530E+00   -.9647E+01
THREE-POINT INTERPOLATION

PROPOSED DESIGN
ALPHA = .10830E+00
X-VECTOR     .1988E+00   .5879E+02   .4974E-02   .756.E+01
OBJ = -.11352E+05
CONSTRAINT VALUES
-.3796E+00   -.7204E+00   -.4476E+00   -.3552E+01   -.1151E+00   -.9605E+01
** END OF ONE-DIMENSIONAL SEARCH
CALCULATED ALPHA = .10330E+00
OBJ = -.113519E+05
DECISION VARIABLES (X-VECTOR)
1) .46298E+02   .19375E+00   .58786E+02   .49743E-02   .75680E+01
CONSTRAINT VALUES (G-VECTOR)
1) -.37958E+00   -.72042E+00   -.44764E+00   -.35524E+01   -.11509E+00   -.93349E+01

BEGIN ITERATION NUMBER 3
CT = -.5000E-01   CTL = -.1000E-01   PHI = .5000E+02
THERE ARE 0 ACTIVE CONSTRAINTS
THERE ARE 0 VIOLATED CONSTRAINTS
THERE ARE 0 ACTIVE SIDE CONSTRAINTS
GRADIENT OF OBJ
1) .21386E+04   .65462E+03   -.12761E+04   .34363E+03   .12502E+05
SEARCH DIRECTION (S-VECTOR)
1) -.17106E+00   -.62361E-01   .10207E+00   -.27435E-01   -.10000E+01
ONE-DIMENSIONAL SEARCH
INITIAL SLOPE = -.1304E+05 PROPOSED ALPHA = .6103E-01

** CONSTRAINED ONE-DIMENSIONAL SEARCH INFORMATION **

PROPOSED DESIGN
ALPHA = .61089E-01
X-VECTOR     .1991E+00   .5918E+02   .4326E-02   .7073E+01
.4567E+02

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OBJ = -.12200E+05  
CONSTRAINT VALUES  
- .4315E+00 - .6885E+00 - .5004E+00 - .3500E+01 - .2823E+01 - .7177E+01

## TWO-POINT INTERPOLATION

PROPOSED DESIGN  
ALPHA = -.22297E+00  
X-VECTOR  
.4401E+02 .1974E+00 .5992E+02 .4344E-02 .5784E+01

OBJ = -.14254E+05  
CONSTRAINT VALUES  
- .5958E+00 - .5042E+00 - .5057E+00 - .3494E+01 - .1181E+02 .1009E+01

## THREE-POINT INTERPOLATION

PROPOSED DESIGN  
ALPHA = .19387E+00  
X-VECTOR  
.4431E+02 .1377E+00 .5976E+02 .4946E-02 .6017E+01

OBJ = -.13941E+05  
CONSTRAINT VALUES  
- .5617E+00 - .5383E+00 - .5277E+00 - .3472E+01 - .1102E+02 .1924E-01

• • • END OF ONE-DIMENSIONAL SEARCH

CALCULATED ALPHA = .6109E-01

OBJ = -.121995E+05

DECISION VARIABLES (X-VECTOR)  
1) .45671E+02 .19511E+00 .59097E+02 .49659E-02 .70793E+01

CONSTRAINT VALUES (G-VECTOR)  
1) -.4314E+00 -.66854E+00 -.50044E+00 -.34596E+01 -.28234E+01 -.71766E+01

BEGIN ITERATION NUMBER 4

CT = -.50000E-01 CTL = -.10000E-01 PH = .50000E+02

NEW SCALING VECTOR (SQRL)  
.4567E+02 .1991E+00 .5310E+02 .4966E-02 .7079E+01

THERE ARE 0 ACTIVE CONSTRAINTS

THERE ARE 0 VIOLATED CONSTRAINTS

THERE ARE 0 ACTIVE SIDE CONSTRAINTS

GRADIENT OF OBJ  
1) .21913E+04 .76217E+03 -.15347E+04 .31350E+03 .15512E+05

SEARCH DIRECTION (S-VECTOR)  
1) -.18542E+00 -.64550E-01 .13029E+00 -.26561E-01 -.10000E+01

ONE-DIMENSIONAL SEARCH  
INITIAL SLOPE = -.1245E+05 PROPOSED ALPHA = .6514E-01

• • CONSTRAINED ONE-DIMENSIONAL SEARCH INFORMATION • •

PROPOSED DESIGN  
ALPHA = .85137E-01  
X-VECTOR  
.4495E+02 .1380E+00 .5975E+02 .4355E-02 .6477E+01  
OBJ = -.13251E+05  
CONSTRAINT VALUES  
-5037E+00 -5963E+00 -5391E+01 -3461E+01 -6604E+01 -3336E+01

TWO-POINT INTERPOLATION

PROPOSED DESIGN  
ALPHA = .16161E+00  
X-VECTOR  
.4430E+02 .1970E+00 .6034E+02 .4845E-02 .5935E+01  
OBJ = -.14001E+05  
CONSTRAINT VALUES  
-5735E+00 -5265E+00 -5265E+00 -3473E+01 -1030E+02 .3603E+00

THREE-POINT INTERPOLATION

PROPOSED DESIGN  
ALPHA = .15455E+00  
X-VECTOR  
.4436E+02 .1971E+00 .6029E+02 .4346E-02 .5685E+01  
OBJ = -.14006E+05  
CONSTRAINT VALUES  
-5663E+00 -5337E+00 -5302E+00 -3470E+01 -1000E+02 .2677E-02

• • END OF ONE-DIMENSIONAL SEARCH

CALCULATED ALPHA = .15455E+00  
OBJ = -.140057E+05  
DECISION VARIABLES (X-VECTOR)  
1) .44362E+02 .19712E+00 .60277E+02 .43455E-02 .53352E+01  
CONSTRAINT VALUES (G-VECTOR)  
1) -.56625E+00 -.53375E+00 -.53022E+00 -.34695E+01 -.10001E+02 .26773E-02

BESTN ITERATION NUMBER 5

CT = -.50000E-01 (TL = -.10000E-01) +MI = .50000E+02

THERE ARE 1 ACTIVE CONSTRAINTS

CONSTRAINT NUMBERS ARE  
6

THERE ARE 0 VIOLATED CONSTRAINTS

THERE ARE 0 ACTIVE SIDE CONSTRAINTS

GRADIENT OF OBJ

1) .23161E+04 .88261E+01 -.23661E+04 .13566E+03 .10241E+05

GRADIENTS OF ACTIVE AND VIOLATED CONSTRAINTS

CONSTRAINT NUMBER 6

1) -.538E+02 -.66882E+01 -.4677E+02 -.31770E+01 -.69400E+02

PUSH-OUT FACTORS, (THETA1), (1,MAC)

1) .11108E+01

CONSTRAINT PARAMETERS, PETA = .71116E+00

SEARCH DIRECTION (S-VECTOR)

1) -.10000E+01 -.17270E+01 .9119E+00 .24147E-01 -.30729E+00

ONE-DIMENSIONAL SEARCH

INITIAL SLOPE = -.7650E+04 PROPOSED ALPHA = .1865E+00

• • CONSTRAINED ONE-DIMENSIONAL SEARCH INFORMATION • •

PROPOSED DESIGN

ALPHA = .13534E+00

X-VECTOR

.3611E+02 .1967E+00 .678E+02 .4362E-02 .5637E+01

OBJ = -.15027E+05

CONSTRAINT VALUES

-.6106E+00 -.8994E+00 -.5146E+00 -.1485E+01 .2546E+00 -.1025E+02

TWO-POINT INTERPOLATION

PROPOSED DESIGN

ALPHA = .13354E+00

X-VECTOR

.3826E+02 .1967E+00 .6764E+02 .4962E-02 .5635E+01

OBJ = -.15004E+05

CONSTRAINT VALUES

-.6095E+00 -.4906E+00 -.5152E+00 -.1483E+01 .1411E-01 -.1061E+02

THREE-POINT INTERPOLATION

PROPOSED DESIGN

ALPHA = .13334E+00

X-VECTOR

.3827E+02 .1967E+00 .6763E+02 .4962E-02 .5655E+01

OBJ = -.15006E+05

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CONSTRAINT VALUES  
 -.6094E+00 -.4906E+00 -.5153E+00 -.3485E+01 -.7405E-02 -.1001E+02  
 \* \* \* END OF ONE-DIMENSIONAL SEARCH  
 CALCULATED ALPHA = 0.  
 OBJ = -.140057E+05 NO CHANGE ON OBJ  
 DECISION VARIABLES (X-VECTOR)  
 1) .44362E+02 .13712E+00 .60287E+02 .49455E-02 .59852E+01  
 CONSTRAINT VALUES (G-VECTOR)  
 1) -.56625E+00 -.53375E+00 -.53022E+00 -.14630E+01 -.10003E+02 .26773E-02

FINAL OPTIMIZATION INFORMATION

OBJ = -.140057E+05  
 DECISION VARIABLES (X-VECTOR)  
 1) .44362E+02 .13712E+00 .60287E+02 .49455E-02 .59852E+01  
 CONSTRAINT VALUES (G-VECTOR)  
 1) -.56625E+00 -.53375E+00 -.53022E+00 -.14630E+01 -.10003E+02 .26773E-02

THERE ARE 1 ACTIVE CONSTRAINTS  
 CONSTRAINT NUMBERS ARE  
 5

THERE ARE 0 VIOLATED CONSTRAINTS

THERE ARE 0 ACTIVE SIDE CONSTRAINTS

TERMINATION CRITERION  
 ITER EQUALS ITMAX

NUMBER OF ITERATIONS = 5

OBJECTIVE FUNCTION WAS EVALUATED 41 TIMES

CONSTRAINT FUNCTION WERE EVALUATED 41 TIMES

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\*\*\*\*\*FLAME RESULTS\*\*\*\*\*

THE FUEL IS CO  
THE OXIDIZER IS N2 O

CHI = 0.0000 Q = 0.0000 P = 5.9052 S = 0.0000 ET-N2 = 1.0000

\*\*REACTANTS\*\*

	MASS	MOLE
CARBON	0.0000	0.0000
CHON	.1169	.1252
NI (O <sub>2</sub> )	.1836	.1252
N2	.6995	.7455
CO2	0.0000	0.0000

\*\*PRODUCTS\*\*

	MASS	MOLE
CO2	.1836	.1252
H2O	0.0000	0.0000
O2	0.0000	0.0000
N2	.6164	.3745

FUEL-AIR RATIO = .1223 EXCESS AIR = 0.0000

FLAME TEMP = 1556.2503 DEG. K

DELTA = 0.000 DEG. K

DELTA2 = 0.000 DEG. K

0.

BTU/MOLE

BTU/MOLE FOR 1.00 OF THE TOTAL NITROGEN

ADDITIONAL N2O ENTHALPY = 0.

ADDITIONAL ENTHALPY = 0.



```
*****AVUL0 GAIN MODEL, NASA/AMES VERSION 2.0*****
      GCL OPTIMIZATION SAMPLE
      OPTIMUM GCL
      N20 CYCLES
```

NOZZLE THROAT HEIGHT (MM) =  
NOZPINAL RATE FOR PC2H

```

TSTAG = 1566.25 DEG K      PSTAG = 44.3621 ATM      AEXIT/A* = 60.24
C02 FRACTION = .12523      H2O FRACTION = .00435      C0 FRACTION = .00000
N2 FRACTION = .86982      J = 0      IFAUSB = 0      IASRT = 0      IFLUX = 0
IPLOT = 3      IJ = 0      DELX = 2.0000      ELSUE = .103957 CM
CURVE = 0.000000      NMAX = 100000      LXMUX = .5000000
C = 0.000000      FAD = 0.000000      THETA1 = 0.0000
CM1 = -.1314568E+02      CM2 = .607339E+02      CM3 = .1261430E+03
THE STANDARD DEVIATION FOR THE CMS IS 0.000000      THE NUMBER OF POINTS TAKEN IS 0
FRICTION FACTOR = 0.0000      PROLESS GAMMA = 1.26161      WEIGHTED GAS CONSTANT OF MIXTURE ERG2/GM-JEG K = .27743E+07
SWITCHOVER AT M = 1.050000      A/L* = 1.002024      X = .00002 CM
B1 = .100E+01      B2 = .500E+00      B3 = .100E+01      B4 = .143E+00      B5 = .100E+01
B7 = .100E+01      B8 = .10E+01      B10 = .100E+01      B11 = .100E+01      B12 = .100E+01
B19 = .100E+01      B19 = .100E+01      B19 = .100E+01

```

TRANSITION LIFETIME = .536E+01 SEC

OPTICAL CROSS SECTIONS =  $\cdot 130E-13$  CM<sup>2</sup> FOR CO<sub>2</sub>,  
=  $\cdot 670E-14$  CM<sup>2</sup> FOR N<sub>2</sub>,  
=  $\cdot 380E-14$  CM<sup>2</sup> FOR H<sub>2</sub>O,  
=  $\cdot 370E-14$  CM<sup>2</sup> FOR HE

[illegible]



X (CM)	M	AV*	PISTOL	T (DEG K)	T01 (DEG K)	I C2 (DEG K)	I C3 (DEG K)	IN (DEG K)	TH2 (DEG K)
DX	DM/LX	O/D (P/B)	CF/CH	I /IX	DTG/CH	DTG/CH	DTG/CH	DTG/CH	DTG/CH
UICH/SEC	EC1/RIDEK K	EC2/RIDEK K	EC3/RIDEK K	EC4/RIDEK K	EN/RIDEK K	EN2/RIDEK K	EN3/RIDEK K	EN4/RIDEK K	EN5/RIDEK K
1/LC2N	1/LC2N	1/LC2N	1/LC2N	1/LC2N	1/LC2N	1/LC2N	1/LC2N	1/LC2N	1/LC2N
1/LC32E	WCT/RIDEK K	WCT/RIDEK K	WCT/RIDEK K	WCT/RIDEK K	WCT/RIDEK K	WCT/RIDEK K	WCT/RIDEK K	WCT/RIDEK K	WCT/RIDEK K
1/LNHE	FLJPI (W/C)	FLJPI (W/C)	FLJPI (W/C)	FLJPI (W/C)	FLJPI (W/C)	FLJPI (W/C)	FLJPI (W/C)	FLJPI (W/C)	FLJPI (W/C)
GAINJP (1/CM)	GAINJP (1/CM)	GAINJP (1/CM)	GAINJP (1/CM)	GAINJP (1/CM)	GAINJP (1/CM)	GAINJP (1/CM)	GAINJP (1/CM)	GAINJP (1/CM)	GAINJP (1/CM)
41279E+01	52386E+01	60284E+02	3249E+01	2627E+03	5600E+03	5600E+03	13014E+04	13551E+04	55405E+03
14603E+00	4824E+01	4224E+01	7946E+03	8407E+01	5617E+02	5617E+02	1794E+01	40214E+01	42036E+01
907	1671E+06	8662E+01	5242E+02	3401E+02	3623E+03	3623E+03	3372E+02	1637E+02	3924E+11
5398E-12	1119E-01	2968E-01	6307E+02	3304E+01	2220E-02	2220E-02	2510E-01	2193E+01	2652E+07
19835E-03	5546E+05	8318E+04	1507E+04	1035E-17	1534E+02	1534E+02	5300E-09	4224E+00	4525E-04
40631E-03	40631E-03	16945E+03	8620E-02	1035E+05	15	15	9325E+00	9324E+00	
6103E+01	5130E+01	6027E+02	3333E+01	2722E+03	4664E+03	4664E+03	1377E+04	1375E+04	4796E+03
1485E+00	3862E-01	0	4946E+03	3510E+01	3600E+02	3600E+02	13261E+01	3400E+01	30640E+02
921	1669E+06	3385E+01	3519E+02	3372E+02	2641E+03	2641E+03	3372E+02	3372E+02	6604E-11
9061E-12	1242E-01	2325E-01	6713E+02	3230E+01	7722E-02	7722E-02	2222E-01	2135E+01	2725E+07
2475E+03	2576E+05	6318E+04	1609E+04	3234E-17	1379E-02	1379E-02	5300E-09	5619E+04	
6170E-03	5733E-03	2134E+03	12110E-01	1262E+05	15	15	9326E+00	9325E+00	4530E-04
8281E+01	5068E+01	6027E+02	3466E+01	2795E+03	4063E+03	4063E+03	1234E+04	1340E+04	4256E+03
15011E+00	2301E-01	0	3627E-03	3350E+01	2234E+02	2234E+02	2305E+01	3036E+01	21684E+02
935	1669E+06	2152E+01	2504E+02	3343E+02	2606E+02	2606E+02	3130E+03	3347E+01	6621E-11
1238E-11	1322E-01	2760E-01	6616E+02	3193E+01	7135E-02	7135E-02	2267E-01	2150E+01	2775E+07
2742E+03	2598E+05	6318E+04	1636E+04	5521E-17	1212E+02	1212E+02	9399E+260	5545E+04	
7103E-03	6612E+03	2304E+03	1402E-01	1402E+05	15	15	9327E+00	9326E+00	4531E-04
1070E+02	5029E+01	6027E+02	3513E+01	2835E+03	3695E+03	3695E+03	1268E+04	1347E+04	3974E+03
15110E+00	1466E-01	0	1548E-03	1561E+01	1320E+02	1320E+02	2237E+01	2635E+01	1440E+02
949	1669E+06	1930E+01	1932E+02	3744E+02	2573E+02	2573E+02	3120E+03	3347E+01	1036E-10
1492E-11	1375E-01	2575E-01	6616E+02	3166E+02	3469E-02	3469E-02	2356E-01	2170E+01	2600E+07
2953E+03	2613E+05	8318E+04	1640E+04	5700E-17	1063E+02	1063E+02	9399E+260	4747E+04	
7526E-03	7006E-03	2304E+03	1479E-01	1402E+05	15	15	9327E+00	9326E+00	4535E-04
1250E+02	5003E+01	6027E+02	3538E+01	2862E+03	3461E+03	3461E+03	1268E+04	1320E+04	3630E+03
1517E+00	1023E-01	0	1414E-03	1400E+01	2542E+03	2542E+03	3033E+03	2730E+01	6624E-10
963	1667E+06	5717E+00	1679E+02	3274E+02	2028E+01	2028E+01	3243E+03	3243E+03	1224E-10
1648E-11	1412E-01	26450E-01	6546E+02	3142E+01	3663E-02	3663E-02	2150E-01	2170E+01	2820E+07
3106E+03	2624E+05	8318E+04	1620E+04	5642E-17	2028E+01	2028E+01	5300E-09	4066E+04	
7679E-03	7144E-03	2304E+03	1524E+05	1426E+05	15	15	9327E+00	9326E+00	4536E-04
1485E+02	4942E+01	6027E+02	3502E+01	2811E+03	3367E+03	3367E+03	12791E+04	1320E+04	3495E+03
1528E+00	7842E-02	0	1024E-03	6545E+02	3879E+01	3879E+01	2223E+01	2669E+01	4533E+03
977	1667E+06	4021E+00	1473E+02	3243E+02	2513E+03	2513E+03	2322E+03	2669E+01	4533E+03
1854E-11	1440E-01	2697E-01	6566E+02	3166E+02	1601E-01	1601E-01	2322E+03	2669E+01	4533E+03
3215E+03	2632E+05	8318E+04	1700E+04	6002E-17	7897E+01	7897E+01	2322E+03	2669E+01	4533E+03
7670E-03	7154E-03	2367E+03	15702E-01	1405E+05	15	15	9327E+00	9326E+00	4537E-04
1677E+02	4563E+01	6027E+02	3501E+01	2811E+03	3367E+03	3367E+03	12791E+04	1320E+04	3495E+03
1526E+00	6640E-02	0	9567E-04	7323E+02	3305E+03	3305E+03	1741E+04	1374E+04	3423E+03
991	1667E+06	7237E+00	1473E+02	3243E+02	2513E+03	2513E+03	2322E+03	2669E+01	4533E+03
1955E-11	1464E-01	2741E-01	6566E+02	3166E+02	1601E-01	1601E-01	2322E+03	2669E+01	4533E+03
3111E+03	2639E+05	8318E+04	1715E+04	6154E-17	7897E+01	7897E+01	2322E+03	2669E+01	4533E+03
7671E-03	7092E-03	2347E+03	15714E-01	1405E+05	15	15	9327E+00	9326E+00	4537E-04

[illegible]

.....  
.....

..... OPTIMUM GOL  
..... N2O OXIDIZER

# GOL OPTIMIZATION SAMPLE

THE MAXIMUM GAIN IS .15302E+01 PCT/CM THIS OCCURS AT X = 14.83736 CM  
THE 1/2 OF THE MAX GAIN IS .7651E+00 PCT/CM THE OCCURS AT X = 3.74725 CM

THERE IS NO VALUE OF GAIN SUITABLE FOR THE REQUIRED CONDUCTION

THE NUMBER CHARACTERIZING THE TAILING PORTION OF THE CURVE IS .5552042

THE NUMBER CHARACTERIZING THE DUTY OF THE TAILING PORTION OF THE CURVE IS

THE NUMBER AVERAGE SLOPE IS -.00932

.5102159

THE AVERAGE SPECIFIC POWER OVER A 10.000 CM DISTANCE IS 14005.6771 J/LC.  
IT WAS AVERAGED BETWEEN 8.241 CM AND 18.241 CM.

ORIGINAL PAGE IS  
OF POOR QUALITY

***WEIGHT/VOLUME RESULTS***		STAGNATION TEMPERATURE (DEGREES KELVIN)		1566.25	
TOTAL PRESSURE (PSI)	552.12	SPECIFIC POWER (JOULES/LEW) (EFF = 1.00)	14005.68		
TOTAL POWER (WATTS)	50000.00	SHOT LENGTH (SECONDS/FIRING)	2.08		
NUMBER OF SHOTS	15	MASS FLOW RATE (LBM/SECOND)	35.78		
TOTAL FLOW TIME (SECONDS)	48.00	LASER WIDTH (INCHES)	73.66		
LASER AND MIRROR HEIGHT (INCHES)	5.00	NOZZLE BODILY LAYER FACIOP	555426E-05		
CAVITY ASPECT RATIO	14.73	REQUIRED DIFFUSER ENTRANCE PRESSURE (PSI)	552.10		
DIFFUSER LENGTH (INCHES)	7.22	DIFFUSER EXIT PRESSURE (PSI)	14.70		
DIFFUSER EXIT MACH NUMBER	.20				
SYSTEM AND SUBSYSTEM HEIGHTS AND VOLUMES		HEIGHT (LB)		VOLUME (FT**3)	
*COMBUSTOR		73.34	3.675		
*CAVITY, NOZZLE, MANIFOLD		361.43	5.371		
SIDEWALL		116.13			
NOZZLE		171.73			
ENDWALL		60.00			
MANIFOLD		13.63			
*STRUCTURE		107.62			
*MOUNTS		43.11			
*DIFFUSER SET- = .00		550.21	1.541		
VANES ( 55)		149.13			
SIDEWALLS		31.33			
ENDWALLS		356.93			
MOUNTS		12.32			
*OPTICAL BENCH		39.51	10.140		
*ADVANCED OPTICAL BENCH		500.00	10.000		
*FUEL SUPPLY SYSTEM		1403.04	45.532		
FUEL		1478.40	23.563		
TANK		44.35			
PUMP		49.25	2.456		
VALVE		71.04	23.570		
*MISCELLANEOUS		351.00	16.000		
TOTAL		3215.42	122.453		
TOTAL USING A.V. M.L. BENCH		3675.51	122.313		

Figure 3.- Concluded

## REFERENCES

1. Otten, L. J., III; Saunders, R. C., III; and Morris, S. J.: System Optimization of Gasdynamic Lasers. NASA TM X-73,192, 1976.
2. Saunders, R. C., III; and Otten, L. J., III: Proceedings of the Symposium on Gasdynamic and Chemical Lasers, Cologne, Oct. 1976.



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16. Abstract  This report contains the user's manual for a computer program that performs system optimization of gasdynamic lasers. Detailed input/output formats are CDC 7600/6600 computers using a dialect of FORTRAN. The manual is intended to be used in conjunction with NASA TM-73,192, "System Optimization of Gasdynamic Lasers," which contains the program structure. Sample input/output data are provided to verify correct program operation along with a program listing.					
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